

### Hydrodynamics of Shocked Interfaces



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### Summary



- 1. Shock waves occur in a wide range of engineering and scientific contexts including laser experiments, astrophysics, and inertial confinement fusion (ICF).
- 2. High-powered lasers can drive strong waves in materials, enabling their study in extreme pressure environments. We developed a method for further strengthening such shocks.
- 3. Shocked interfaces mix via the Richtmyer-Meshkov instability (RMI), possibly leading to the escape of high-vorticity ejecta with important implications for turbulent transitions.



# Shock waves occur in a variety of engineering and natural contexts where they interact with fluid interfaces.



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#### Pressure



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Pressure



An intermediate material bridging an impedance discontinuity can transiently strengthen a shock wave in a material of interest.







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A single intermediate material can increase pressures by up to 16%, as verified by HYADES simulations utilizing tabular equations of state.





[8]

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Energy and circulation deposited when a shock wave passes through a perturbed interface leads to significant interfacial mixing.





Shock impinges on heavy-to-light interface

Shock deposits baroclinic vorticity

Rarefaction reflects and interface evolves



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М  $ho_L$  $\rho_H$  $\nabla p$  $\nabla p$ 

Shock impinges on heavy-to-light interface

Shock deposits baroclinic vorticity

Rarefaction reflects and interface evolves





Linear theory and nonlinear models predict the width of the mixing layer for multimode interfaces, with an eventual handoff to turbulence models.



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The transition to turbulence of the RMI is not well understood and involves several complicating transport mechanisms.









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[2]

The transition to turbulence of the RMI is not well understood and involves several complicating transport mechanisms.



[11]



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In addition to the RMI, shocked interfaces can generate jet/ejecta flows with important applications in ICF, ejecta physics, and astrophysical jets.



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Classical studies focus on vortex rings generated in a piston-cylinder apparatus and have uncovered key insights into their scaling behavior.





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[19]





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[19]



Piston displacement, L/D

The formation number is based on analysis comparing the amount of energy supplied by the piston to the energy that can reside in the ring.





<b>Slug Flow</b>	
$E = \frac{1}{8}\pi D^2 L U_p^2$	
$\Gamma = \frac{1}{2}LU_p$	
$I = \frac{1}{4}\pi D^2 L U_p$	
$U_{tr} = \frac{\partial E}{\partial I} = \frac{1}{2} U_p$	

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![](_page_26_Picture_1.jpeg)

![](_page_26_Figure_2.jpeg)

We modify the classical theory for vortex ring scaling by considering the strength of the shock and the interfacial Atwood number.

![](_page_27_Figure_1.jpeg)

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![](_page_27_Picture_4.jpeg)

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We modify the classical theory for vortex ring scaling by considering the strength of the shock and the interfacial Atwood number.

![](_page_28_Picture_1.jpeg)

![](_page_28_Figure_2.jpeg)

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# The model predicts an augmented formation number, and simulations show many formation number signatures.

![](_page_29_Picture_1.jpeg)

![](_page_29_Figure_2.jpeg)

![](_page_29_Figure_3.jpeg)

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## Simulations verify that our model captures the augmented formation number scaling of vortex rings ejected from shock-accelerated interfaces.

![](_page_30_Picture_1.jpeg)

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### Summary

![](_page_31_Picture_1.jpeg)

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![](_page_31_Picture_5.jpeg)

![](_page_31_Picture_6.jpeg)

![](_page_31_Picture_7.jpeg)

![](_page_31_Picture_8.jpeg)

### References

![](_page_32_Picture_1.jpeg)

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